1 Modal analysis of offshore monopile wind turbine: An analytical solution

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24 ABSTRACT

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26 An analytical solution of the dynamic response of offshore wind turbines under wave load with 27 nonlinear Stokes's wave theory and wave-structure and soil-foundation interactions is developed. 28 Natural frequencies and the corresponding modes are obtained. The effect of the wave-structure 29 interaction, the added mass, the foundation stiffness, and the nacelle translational and rotational 30 inertia on the motion of the structure is investigated. The nonlinear loading provided by the drag 31 term of Morison's equation is successfully handled. A parametric study to examine the effect of 32 the structural parameters on the dynamic response is conducted and the results of the proposed 33 analytical solution are compared to numerical ones. The proposed method has the following 34 advantages: a) it is accurate and straightforward because of its analytical nature, b) it does not 35 ignore the drag term in the wave loading by keeping its nonlinearity nature, c) the structure of 36 the wind turbine is modeled as a continuous system, d) it takes into account the effect of the 37 rotational and translational inertia of the nacelle on the dynamic response, e) it provides an 38 interpretation of the effect of the sea level variation in changing the natural frequencies. 39

- 40 Keywords: Offshore Wind Turbine, Response, Natural frequencies, Natural modes, Wave-
- 41 Structure interactions, Nonlinear wave kinematics, Soil-Structure Interactions.
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43 1. Introduction

Passing from carbon-emission to carbon-neutral energy sources motivates many countries to invest in renewable energies. Among them, offshore wind turbines (OWTs) have been chosen as the main source of renewable energy for many European countries [1]. Being installed in the offshore and nearshore mostly by fixed support structures poses challenges in optimum designing. Furthermore, the prediction of the long-term structural behavior of these structures required for structural integrity assessment is not available because the majority of the OWTs have been installed in the past ten years, and the data acquisition is very costly.

51 OWTs are subjected to coupled dynamic phenomena due to the interaction of wind and wave 52 loads and the rotor's vibration. Therefore, dynamic stability and vibration control [2,3] is a topic 53 that attracts the attention of researchers. Pioneering works on vibrational control of large real-54 life structures under dynamic loading have been performed by Adeli and Saleh [4–7]. Reviews of 55 advances in vibration control algorithms for smart structures up to 2017 are presented by [8-10]. 56 Since the coupling phenomena are complicated to be modeled by regular deterministic 57 techniques, Machine Learning (ML) algorithms seem to be a promising alternative tool. Adeli and 58 associates developed intelligent control algorithms employing neural networks and ML 59 techniques as far back as 2008 [11,12]. More recent work on intelligent control of large real-life 60 buildings and bridge structures was presented by [9,13–19] to discuss and advance the novel 61 concept of integration of vibration control, health monitoring of structures [20,21] and energy 62 harvesting [22] for smart cities of the future.

63 A key technology in the field of structural engineering in recent years has been automated 64 structural health monitoring (SHM). There are two fundamentally different approaches to SHM, 65 one based on vibration [23] and the other based on imaging and computer vision [24]. Vibration-66 based SHM technology requires adroit integration of vibration theory, signal processing such as 67 wavelets [25], and machine learning [26,27]. SHM technology has been used successfully and there is a significant body of research on health monitoring of building structures and bridges [28], 68 69 dams [29], railways [30], pavements [31], retaining structures along highways [32], and tunnels [33], but little work has been reported on health monitoring of offshore structures and wind 70 71 turbines. A method of vibration-based SHM is based on computing and monitoring the structural 72 properties such as natural frequencies and mode shapes [34]. Besides, the remaining fatigue life 73 is also necessary to be evaluated and monitored in the case of offshore structures. One of the 74 main challenges in fatigue life estimation, apart from selecting a novel accumulating damage 75 model [35–37], is the availability of the stress history resulting from dynamic response of the 76 structure in the hot spots.

The stress history required for fatigue life evaluation can be obtained from two main sources. One source would be simply by measuring the stress history of the real operating structure. Although the field data is extremely valuable because it reflects the real behavior of the structure, harvesting field data is expensive and in some cases, impossible. An alternative source would be generating data through the mathematical models by simulating the real operating situation. An immediate method in providing the response or stress history of an OWT structure, including all its complexities, is to utilize the numerical models by discretizing the structure via finite elements

84 methods. Many researchers [38–48] are using commercial software packages or numerical 85 platforms to simulate these structures due to their accessibility and ability to create high-fidelity 86 models. However, numerical solutions are still time-consuming, and their performance is affected 87 by stability issues especially in very complicated time domain problems. The other alternative to 88 generating response data is to develop the analytical solutions to achieve the response as a single 89 function by which the response is evaluated for arbitrary loading parameters. Analytical solutions 90 are straightforward, provide always-true and reliable results as opposed to the numerical ones 91 but challenging to achieve because of complex mathematics.

92 Unlike a large number of existing numerical models, very few analytical models of the OWTs have 93 been published. Scientists such as Graff [49] and Meirovitch [50] started early works on the 94 classical methods of solving the wave equation in a cantilever flexural beam in the 60s and 70s. 95 They introduced some methods for solving these equations. For instance, Graff [49] listed five 96 methods: 1) Finite Fourier transform, 2) Expansion in the natural modes (in the spatial domain), 97 3) Laplace transform, 4) Laplace transform-natural mode expansion, and 5) solution by the natural 98 modes (in both spatial and temporal domain). In the first two methods, it is assumed that the 99 solution contains two separate parts in time and space. While for the last three ones, the solution 100 begins directly from the equation of motion with an arbitrary function. Among these methods, 101 expansion in the natural modes relies on the structure's natural modes, which can be found from 102 the homogeneous form of the equation of motion with respect to its boundary conditions. This 103 method also provides the result in the form of a single function as apposed to numerical solutions 104 where a set of numbers is obtained as the response. Expansion in the normal modes is also flexible 105 in the external load situations. Complex loading can be expanded by using the Fourier series and 106 analyzed. Selecting other methods requires more effort, and sometimes impossible to find the 107 solution. For instance, the method of Laplace transforms requires inverse Laplace transform, 108 which is nearly impossible to be determined analytically for complicated boundary conditions. 109 Pavlou [51] developed an analytical solution for the evaluation of the response of the OWT under 110 the linear waves. In this work, the translational and rotational inertial effect of the nacelle, the 111 hydrodynamic damping, and the soil-foundation interaction have been analytically investigated 112 for gravity-based supported structures. The achieved analytic inversion of Laplace Transforms was 113 very challenging in this analysis. Apart from mentioned methods, very few analytical solutions 114 have been proposed in the past years. In a rare case, Wang et al. [52] developed a mathematical 115 model for dynamic analysis of an onshore wind turbine by using the thin-walled theory [45,53– 116 55] to simulate the comprehensive behavior of the wind turbine.

117 Expansion of the response in the natural modes requires having an accurate and reliable 118 estimation of the natural frequencies of the OWT structure. In the past few years, researchers 119 have attempted to provide models in which the realistic situation of an OWT is included. Most of 120 the works have been focused on the soil and foundation situation. In a study conducted by Arany 121 et al. [56], an analytical model was developed by simulating foundation flexibility using three 122 springs. The effect of boundary conditions on the natural frequencies has been parametrically 123 studied by defining some non-dimensional parameters. Having compared the analytical natural 124 frequencies with the measured ones from the actual OWTs, a slight inaccuracy was reported, 125 which is not improved by modeling the tower with the Timoshenko beam theory. In another 126 study, They [57] proposed a simplified methodology to have a quick hand calculation of the 1st 127 natural frequency of an OWT. In their works, the effect of the fluid-structure interactions of the 128 added mass has not been considered, which may be the reason for the inaccuracy they have 129 reported. Bouzid et al. [58] established a nonlinear finite element model [59] to obtain the head 130 stiffness of the monopile support structure at the mudline. Their purpose was to improve the 131 accuracy of the results obtained by Arany et al. [56,57]. Recently, Alkhoury et al. [60] established 132 a full 3D model for the DTU 10 MW OWT including all the details of the nacelle, blades, rotor, as 133 well as full 3D modeling of the soil inside and outside of the monopile and cross-sectional variation 134 of the tower structure. More details about this work will be presented later in this paper for the 135 sake of verification.

136 The natural frequencies of the OWT were also measured in the real-operating systems. Damgaard 137 et al. [61] have reported the cross-wind modal properties of an OWT. They have reported that the 138 1st natural frequency is time-dependent which might be because of erosion of the soil around the 139 monopile or soil scouring. Later, variation of the natural frequency in time attracts the attention 140 of Prendergat et al. [62,63], resulting in two publications. In their first work [62], the scouring 141 effect on the natural frequency was investigated without considering the effect of added mass in 142 the system. In their second work [63], however, they considered other factors such as water 143 added mass influencing the dynamic properties of the system. Another investigation on the 144 measured data conducted by Dong et al. [64] also reported the time-dependent dynamic 145 properties of an OWT. Moreover, a precise analysis was performed on the measured data by 146 Cosgriff et al. [65]. The 1st natural frequency was separately plotted versus the wave height and 147 wind speed based on 20 min measured data in the calm sea condition while the wind velocity was 148 under the cut in speed. The trend reveals that the 1st natural frequency is reduced as the wave 149 height and wind speed increase. This phenomenon raises suspicion about the effect of added 150 mass on the natural frequencies by sea level variation.

151 Attempts toward providing an accurate response require accurate and realistic inputs in the 152 analysis alongside accurate and realistic models and solutions. In the case of bottom-fixed OWTs 153 which are mostly installed in shallow to intermediate water depth, measurements and studies 154 reveal that nonlinear wave theory should be implemented to simulate the realistic sea states. A 155 study conducted by Natarajan [66] showed that using the 2nd order wave theory significantly 156 increases the extreme loading on the monopile support structure of OWTs. Yingguang Wang [67] 157 utilized a transformed linear method to simulate the 2nd order irregular wave to obtain the wave 158 load by including the sea bottom effect. Shaofeng Wang et al. [68] conducted a case study to 159 investigate the ultimate wave loads on a 10 MW OWT. All of them have reported a significant 160 increase in wave load on the structure.

161 In the present work, a new analytical solution for modal analysis of OWT structures is presented. Nonlinear waves and wave-structure and soil-foundation interactions are accounted for in the 162 solution. The consideration of the 2nd nonlinear wave kinematics improves the ability of the 163 164 method to cover a higher range of wavelength and height, providing more realistic loading on the 165 structure. OWTs are subjected to different types of environmental loading such as wind, waves, 166 ocean currents, earthquakes, ship collisions, etc [69]. Among them, dealing with the wave loads 167 is still a challenging task due to its complexities and uncertainties [70]. Therefore, this study 168 focuses on the wave load. Moreover, the analytical modeling of the translational and rotational 169 inertia effect of the nacelle and the fluid-structure and soil foundation interaction improves the

170 reliability of the dynamic simulation. The novelty of the presented work beyond the published 171 literature is based on the following advantages: The solution is accurate and straightforward 172 because of its analytical nature, it does not ignore the drag term in the wave loading, the structure 173 of the wind turbine is modeled as a continuous system by including its geometrical discontinuities, 174 it takes into account the effect of the rotational and translational inertia of the nacelle, and it 175 provides an interpretation of the effect of the sea level variation on the natural frequencies. The 176 research presented in this paper can be extended for health monitoring of offshore structures 177 and wind turbines which is intended as part of future research by the authors.

178 2. Formulation of the problem

179 A typical horizontal axis OWT consists of nacelle and blades systems mounted on the top of a 180 tower fixed to the seabed by a monopile, as illustrated in Figure 1-a. A transition part connects 181 the tower and monopile at sea level. In this paper, an OWT is modeled as a cantilever column 182 supported by a set of springs at one end and free at the other end, see Figure 1-b. This cantilever 183 beam, which is called the system for the rest of this paper, consists of two parts separated from 184 the platform level at the top of the transition part. This is because their dimensions and properties 185 can be significantly different. The cross-sectional properties of each part are assumed constant. 186 The symbols definition used in this paper is represented in Table 1. The monopile under the 187 seabed is also modeled by a set of four springs representing the lateral, rotational, cross-coupled, 188 and vertical stiffnesses with constants of K_L , K_R , K_{LR} , K_z , respectively [71].

189 In this study, the motion of the system is considered as a lateral deflection due to the wave load 190 applied up to Mean Sea Level (MSL). As shown in Figure 1-c, the deflection, x(z, t), is defined as 191 a continuous function of time, t, and space, z, implying to represent the motion of the system with 192 an infinite degree of freedom.

193 2.1. The governing equation

For the system introduced in Figure 1-c, the Bernoulli-Euler beam theory can be applicable for small displacements [49]. Therefore, the equation of motion along the height of the system with the origin from the seabed can be written as

$$EI(z) x^{(iv)}(z,t) + M(z)\ddot{x}(z,t) = q(z,t) H_3(z)$$
(1)

where, q(z,t) is the external forces acting perpendicular to the monopile's longitudinal axis. EI(z) and M(z) are the flexural rigidity and mass per unit length of the system. For the system defined in Figure 1, they are defined as follows:

$$M(z) = m_{Tow} H_1(z) + m_{Mon} H_2(z) + m_{Mon} H_3(z)$$
(2)

$$EI(z) = EI_{Tow}H_1(z) + EI_{Mon}H_2(z) + EI_{Mon}H_3(z)$$
(3)

200 Where $H_i(z)$, i = 1, 2, 3 are the step functions defined as

$$H_{1}(z) = \begin{cases} 1, & for \ L_{Plat} \le z \le L \\ 0, & for \ d < z \le L_{Plat} \\ 0, & for \ 0 < z \le d \end{cases}$$
(4)
$$H_{2}(z) = \begin{cases} 0, & for \ L_{Plat} \le z \le L \\ 1, & for \ d < z \le L_{Plat} \\ 0, & for \ 0 < z \le d \end{cases}$$
(5)
$$H_{3}(z) = \begin{cases} 0, & for \ L_{Plat} \le z \le L \\ 0, & for \ d < z \le L_{Plat} \\ 1, & for \ 0 < z \le d \end{cases}$$
(6)

For a slender structure, Morison's equation from [72,73] can be adopted for dynamic modeling. Also, the relative velocity formulation is applicable for a moving slender structure subjected to the wave loads. Since the bottom-fixed support structure is of interest, it is expected that the underwater motion of the system to be way below its diameter. Therefore, the relative velocity in the drag term can be reduced to the wave horizontal particle velocity. For an ocean wavelength larger than five times the monopile diameter and the small displacement, the wave load on the monopile can be represented by the relative velocity formulation [72] as

$$q(z,t) = -\rho_w C_A A_{Mon} \ddot{x}(z,t) + \rho_w C_M A_{Mon} \frac{Du(z,t)}{Dt} + \frac{1}{2} \rho_w C_D D_{Mon} u(z,t) |u(z,t)|$$
(7)

108 In equation (7), the two first terms inside the curl bracket are inertial and the third one is the drag 209 term. The total derivative of the wave horizontal particle velocity, $Du(z,t)/Dt = \partial u/\partial t + u \partial u/\partial x + w \partial u/\partial z$, in the inertial term can be reduced to $\dot{u}(z,t) = \partial u(z,t)/\partial t$ by neglecting 211 the advocative terms which are reported to slightly increase the load when they are included [70]. 212 The inertia term reveals that the $-\rho_w C_A A_{Mon} \ddot{x}(z,t)$ provides an additional mass to the system 213 affecting its oscillating properties. This added mass can represent itself in the equation of motion,

Eq. (1). By substituting Eq. (7) into Eq. (1) the result will be

$$x^{(iv)}(z,t) + A(z)\ddot{x}(z,t) = \frac{1}{EI_{Mon}}Q(z,t) H_3(z)$$
(8)

215 where

$$A(z) = \frac{1}{a_{Tow}^2} H_1(z) + \frac{1}{a_{MA}^2} H_2(z) + \frac{1}{a_{MU}^2} H_3(z)$$
(9)

$$a_{\rm Tow}^2 = \frac{EI_{Tow}}{m_{Tow}} \tag{10}$$

$$a_{\rm MA}^2 = \frac{EI_{Mon}}{m_{Mon}} \tag{11}$$

$$a_{\rm MU}^2 = \frac{EI_{MON}}{m_{MON} + \rho_w C_A A_{MON}} \tag{12}$$

$$Q(z,t) = \rho_w C_M A_{Mon} \dot{u}(z,t) + \frac{1}{2} \rho_w C_D D_{Mon} u(z,t) |u(z,t)|$$
(13)

Eq. (8) represents a partial differential equation governing the motion of the system subjected to the wave load. The added mass is included on the right side of the equation. Thus, the left side is not dependent on the motion of the tower. Parameter a_{MU} are containing the mass of the system underwater including added mass. Therefore, the wave-structure interaction is included in the equation of motion.

It can be seen in Eq. (8) that there are three separate systems, the tower, monopile above water and monopile underwear, acting together to govern the motion of the system. Therefore, it can be separated in the form of three independent equations as

$$x_{Tow}^{(iv)}(z,t) + \frac{1}{a_{Tow}^2} \ddot{x}_{Tow}(z,t) = 0, \quad for L_{Plat} \le z \le L$$
(14)

$$x_{MA}^{(iv)}(z,t) + \frac{1}{a_{MA}^2} \ddot{x}_{MA}(z,t) = 0, \quad for \ d < z \le L_{Plat}$$
(15)

$$x_{MU}^{(iv)}(z,t) + \frac{1}{a_{MU}^2} \ddot{x}_{MU}(z,t) = \frac{1}{EI_{Mon}} Q(z,t), \quad for \ 0 < z \le d$$
(16)

In the three above equations $x_{Tow}(z, t)$, $x_{MA}(z, t)$ and $x_{MU}(z, t)$ stand for the lateral motion of the system in the tower, monopile above water and underwater, respectively. Eq. (14) and (15) are homogeneous partial differential equations implying that the motion above water is a kind of free vibration whereas the underwater motion, represented in Eq. (16), is a forced vibration due to the external load. Therefore, the above water motion is activated by the motion of the underwater part via a series of boundary conditions which will be introduced later.

230 2.2. The boundary and initial conditions

To accommodate the motion of the system with the equation of motion in Eq. (8), two sets of conditions at the two ends of the system are needed. The motion of the system at the seabed is governed by a set of 4 springs as illustrated in Figure 1-b. Vertical stiffness, K_z , can be neglected because the vertical motion of the system is negligible. The three remaining springs can be collected in the matrix form to obtain the shear force, F, and bending moment, M, at the seabed by the following equation [71]:

$$\begin{bmatrix} F(t) \\ M(t) \end{bmatrix} = \begin{bmatrix} K_L & K_{LR} \\ K_{LR} & K_R \end{bmatrix} \begin{bmatrix} x(0,t) \\ x'(0,t) \end{bmatrix}$$
(17)

By substituting the shear force and bending moment from the beam theory into the aboveequation and expanding, it yields to the following two boundary conditions.

$$-EI_{MON} x_{MU}^{\prime\prime\prime}(0,t) = K_L x_{MU}(0,t) + K_{LR} x_{MU}^{\prime}(0,t)$$
(18)

$$EI_{Mon} x_{MU}^{\prime\prime}(0,t) = K_{LR} x_{MU}(0,t) + K_{R} x_{MU}^{\prime}(0,t)$$
(19)

At the top of the tower, a heavy nacelle is mounted providing lump mass to the system. The effect
of the translational and rotational inertia of the mass of the nacelle can be simulated as the two
following conditions:

$$-EI_{Tow} x_{Tow}^{\prime\prime}(L,t) = J_P \ddot{x}_{Tow}^{\prime}(L,t)$$
⁽²⁰⁾

$$EI_{Tow} x_{Tow}^{\prime\prime\prime}(L,t) = M_N \ddot{x}_{Tow}(L,t)$$
⁽²¹⁾

The rotational motion of the nacelle produces the momentum proportional to the tower's rotational acceleration $\ddot{x}'_{Tow}(L,t)$ at the top of the system which should be in equilibrium with the total moment of the system producing boundary condition in the form of Eq. (20). Besides, the translational acceleration of the nacelle, $\ddot{x}_{Tow}(L,t)$, creates an inertial force that should be equal to the internal shear force of the tower at the nacelle level. This equilibrium is represented by Eq. (21).

As mentioned earlier, the two Eqs. (15) and (16) govern the motion of the monopile. Therefore, it is necessary that these equations are linked together at the sea level via some boundary conditions. They are

$$x_{MU}(d,t) = x_{MA}(d,t)$$
 (22)

$$x'_{MU}(d,t) = x'_{MA}(d,t)$$
 (23)

$$x''_{MU}(d,t) = x''_{MA}(d,t)$$
(24)

$$x_{MU}^{\prime\prime\prime}(d,t) = x_{MA}^{\prime\prime\prime}(d,t)$$
(25)

Also, Eqs. (14) and (15) are connected by following boundary conditions

$$x_{MA}(L_{Plat}, t) = x_{Tow}(L_{Plat}, t)$$
(26)

$$x'_{MA}(L_{Plat},t) = x'_{Tow}(L_{Plat},t)$$
⁽²⁷⁾

$$EI_{Mon}x_{MA}^{\prime\prime}(L_{Plat},t) = EI_{Tow}x_{Tow}^{\prime\prime}(L_{Plat},t)$$
(28)

$$EI_{Mon}x_{MA}^{\prime\prime\prime}(L_{Plat},t) = EI_{Tow}x_{Tow}^{\prime\prime\prime}(L_{Plat},t)$$
⁽²⁹⁾

- 252 The last eight boundary conditions are based on the system's continuity on deflection and slope
- 253 of the motion at sea level as well as the internal shear force and bending moment continuity of
- the system at a point where two systems are linked together.

255 It is assumed that the tower motion starts from the position when it is at the rest. Therefore, the 256 initial deflection and the velocity of each part of the tower are zero. It yields to the following initial

257 conditions

$$x(z,0) = \dot{x}(z,0) = 0$$
(30)

258 **3.** Solution for the equation of motion

- The method that has been chosen to solve Eq. (8) is to expand the response in the natural modesof the system. The natural frequencies of the system and consequently the natural mode shapes
- will be obtained. Then, they will be utilized in the solution for the forced vibration.

262 3.1. Natural modes

The natural modes will be evaluated by using the homogenous form of Eq. (16) as well as Eq. (14)and (15) which are

$$x_{Tn}^{(iv)}(z,t) + \frac{1}{a_{Tow}^2} \ddot{x}_{Tn}(z,t) = 0, \quad for \ L_{Plat} \le z \le L$$
(31)

$$x_{MAn}^{(iv)}(z,t) + \frac{1}{a_{MA}^2} \ddot{x}_{MAn}(z,t) = 0, \quad for \, d < z \le L_{Plat}$$
(32)

$$x_{MUn}^{(iv)}(z,t) + \frac{1}{a_{MU}^2} \ddot{x}_{MUn}(z,t) = 0, \quad for \ 0 < z \le d$$
(33)

265 It should be noted that the subscript n in quantity indicates that it belongs to the n^{th} natural 266 mode. So, $x_{Tn}(z,t)$, $x_{MAn}(z,t)$, and $x_{MUn}(z,t)$ are the natural mode shapes of the tower, 267 monopile above water and monopile underwater, respectively. The solution for the above 268 equations is proposed in the form of

$$x_{Tn}(z,t) = X_{Tn}(z)T_{Tn}(t), \quad for L_{Plat} \le z \le L$$
(34)

$$x_{MAn}(z,t) = X_{MAn}(z)T_{MAn}(t), \quad for \ d < z \le L_{Plat}$$
(35)

$$x_{MUn}(z,t) = X_{MUn}(z)T_{MUn}(t), \quad for \ 0 < z \le d$$
(36)

269 Substituting the Eqs. (34) to (36) into Eqs. (31) to (33) will yield

$$X_{Tn}^{(iv)}(z)T_{Tn}(t) + \frac{1}{a_{Tow}^2} X_{Tn}(z)\ddot{T}_{Tn}(t) = 0, \quad for \ L_{Plat} \le z \le L$$
(37)

$$X_{MAn}^{(iv)}(z)T_{MAn}(t) + \frac{1}{a_{MA}^2} X_{MAn}(z)\ddot{T}_{MAn}(t) = 0, \quad for \ d < z \le L_{Plat}$$
(38)

$$X_{MUn}^{(iv)}(z)T_{MUn}(t) + \frac{1}{a_{MU}^2} X_{MUn}(z)\ddot{T}_{MUn}(t) = 0, \quad for \ 0 < z \le d$$
(39)

270 After some algebraic manipulations, they are transformed to

$$\frac{X_{Tn}^{(iv)}(z)}{X_{Tn}(z)} = \beta_{Tn}^4 = -\frac{1}{a_{Tow}^2} \frac{\ddot{T}_{Tn}(t)}{T_{Tn}(t)}, \quad for \ L_{Plat} \le z \le L$$
(40)

$$\frac{X_{MAn}^{(iv)}(z)}{X_{MAn}(z)} = \beta_{MAn}^4 = -\frac{1}{a_{MA}^2} \frac{\ddot{T}_{MAn}(t)}{T_{MAn}(t)}, \quad for \ d < z \le L_{Plat}$$
(41)

$$\frac{X_{MUn}^{(iv)}(z)}{X_{MUn}(z)} = \beta_{MUn}^4 = -\frac{1}{a_{MU}^2} \frac{\ddot{T}_{MUn}(t)}{T_{MUn}(t)}, \quad for \ 0 < z \le d$$
(42)

- 271 Where β_{Tn} , β_{MAn} and β_{MUn} are the wavenumbers for the tower and monopile above water and
- 272 underwater, respectively. Eqs. (40), (41), and (42) can be separated to form the following
- 273 equations

$$X_{Tn}^{i\nu}(z) - \beta_{Tn}^4 X_{Tn}(z) = 0, \quad for \ L_{Plat} \le z \le L$$
(43)

$$X_{MAn}^{i\nu}(z) - \beta_{MAn}^4 X_{MAn}(z) = 0, \quad for \ d < z \le L_{Plat}$$

$$\tag{44}$$

$$X_{MUn}^{iv}(z) - \beta_{MUn}^4 X_{MUn}(z) = 0, \quad for \ 0 < z \le d$$
(45)

274 And

$$\ddot{T}_{Tn}(t) + a_{Tow}^2 \beta_{Tn}^4 T_{Tn}(t) = 0, \quad for \ L_{Plat} \le z \le L$$
(46)

$$\ddot{T}_{MUn}(t) + a_{MU}^2 \,\beta_{MUn}^4 \,T_{MUn}(t) = 0, \qquad for \, d < z \le L_{Plat} \tag{47}$$

$$\ddot{T}_{MAn}(t) + a_{MA}^2 \,\beta_{MAn}^4 \,T_{MAn}(t) = 0, \qquad for \, 0 < z \le d \tag{48}$$

In the above six equations, the temporal and spatial variables are separated. Therefore, they can be solved independently. Since every section of a continuous system should vibrate with the same natural frequency in each mode shape, so $T_{Tn} = T_{MAn}(t) = T_{MUn}(t)$. By comparing the two Eqs. (46) and (47), one can conclude that $a_{Tow}^2 \beta_{Tn}^4 = a_{MU}^2 \beta_{MUn}^4 = a_{MA}^2 \beta_{MAn}^4$ because $T_{Tn} = T_{MAn}(t) = T_{MUn}(t)$. Therefore,

$$\ddot{T}_n(t) + \omega_n^2 T_n(t) = 0, \quad for \ 0 \le z \le L$$
 (49)

280 Where

$$\omega_n^2 = a_{Tow}^2 \,\beta_{Tn}^4 = a_{MU}^2 \,\beta_{MUn}^4 = a_{MA}^2 \,\beta_{MAn}^4 \tag{50}$$

Since *T* is a periodic function, it will oscillate with the cyclic frequency of ω_n which is a natural frequency of the system.

The solution of the motion Eqs. (34), (35), and (36) can be imposed into the boundary conditions
defined by Eqs. (18) to (29) to yield the boundary conditions independent from the time variable.

285 They are

$$X_{MUn}^{\prime\prime\prime}(0) + \alpha_1 X_{MUn}(0) + \alpha_2 X_{MUn}^{\prime}(0) = 0, \qquad \alpha_1 = \frac{K_L}{EI_{Mon}}, \alpha_2 = \frac{K_{LR}}{EI_{Mon}}$$
(51)

$$X''_{MUn}(0) - \alpha_2 X_{MUn}(0) - \alpha_3 X'_{MUn}(0) = 0, \qquad \alpha_3 = \frac{K_R}{EI_{Mon}}$$
(52)

$$X_{MUn}(d) - X_{MAn}(d) = 0$$

$$X'_{MUn}(d) - X'_{MAn}(d) = 0$$

$$X''_{MUn}(d) - X''_{MAn}(d) = 0$$

$$X''_{MUn}(d) - X''_{MAn}(d) = 0$$

$$X_{MAn}(L_{Plat}) - X_{Tn}(L_{Plat}) = 0$$

$$X'_{MAn}(L_{Plat}) - X'_{Tn}(L_{Plat}) = 0$$

$$X_{MAn}(L_{Plat}) - \alpha_{A}X''_{Tn}(L_{Plat}) = 0$$
(54)

$$X_{MAn}^{\prime\prime}(L_{Plat}) - \alpha_4 X_{Tn}^{\prime\prime}(L_{Plat}) = 0$$

$$X_{MAn}^{\prime\prime\prime}(L_{Plat}) - \alpha_4 X_{Tn}^{\prime\prime\prime}(L_{Plat}) = 0$$

$$EI_{Mon}$$

$$X_{Tn}''(L) - \alpha_5 \omega_n^2 X_{Tn}'(L) = 0, \qquad \alpha_5 = \frac{J_p}{E I_{Tow}}$$
(55)

$$X_{Tn}^{\prime\prime\prime}(L) + \alpha_6 \omega_n^2 X_{Tn}(L) = 0, \qquad \alpha_6 = \frac{M_n}{E I_{Tow}}$$
(56)

- 286 Where α_i , i = 1...6 are the solution variables introduced to implement the soil-structure 287 interactions, nacelle-blades mechanical properties, and the systems mechanical properties at sea 288 level.
- 289 The solution for Eqs. (43), (44) and (45) are in the form of

$$X_{Tn}(z) = T_1 \cos(\beta_{Tn}z) + T_2 \cosh(\beta_{Tn}z) + T_3 \sin(\beta_{Tn}z) + T_4 \sinh(\beta_{Tn}z)$$
(57)

$$X_{MAn}(z) = A_1 \cos(\beta_{MAn}z) + A_2 \cosh(\beta_{MAn}z) + A_3 \sin(\beta_{MAn}z) + A_4 \sinh(\beta_{MAn}z)$$
(58)

$$X_{MUn}(z) = U_1 \cos(\beta_{MUn}z) + U_2 \cosh(\beta_{MUn}z) + U_3 \sin(\beta_{MUn}z) + U_4 \sinh(\beta_{MUn}z)$$
(59)

where T_i , A_i , and U_i are the constant coefficients for each natural mode shape. Substituting the proposed solutions expressed by Eqs. (57), (58), and (59) into the 12 boundary conditions represented by Eqs. (51) to (56) will yield a system of 12 linear equations. In matrix form, they can be represented as

$$\mathbf{P} \times \mathbf{D} = 0, \qquad \mathbf{D} = \{U_1 \ U_2 \ U_3 \ U_4 \ A_1 \ A_2 \ A_3 \ A_4 \ T_1 \ T_2 \ T_3 \ T_4\}^T$$
(60)

- 294 Where **P** is the matrix containing trigonometrical and hyper trigonometrical functions and **D** is the 295 constant coefficients vector. The concept of natural modes oscillation is that the oscillation should 296 be independent of the constant coefficients in Eqs. (57) to (59). Therefore, the determinant of 297 matrix **P** should be zero to yield a singular matrix. In matrix **P**, there are three variables β_{Tn} , β_{MAn}
- 298 , β_{MUn} , and the singularity condition of matrix **P**. Two more conditions are needed to find them
- 299 with one equation. As mentioned earlier, the cyclic frequency of the system is unique so recalling
- 300 the definitions of the wavenumber from Eq. (50) and rewriting them yields

$$\beta_{MAn} = \gamma_{AT} \beta_{Tn}, \qquad \gamma_{AT} = \sqrt[4]{\alpha_4 \frac{m_{Mon}}{m_{Tow}}}$$
(61)

$$\beta_{MUn} = \gamma_{UT} \beta_{Tn}, \qquad \gamma_{UT} = \sqrt[4]{\alpha_4 \left(\frac{m_{Mon} + \rho_w C_A A_{Mon}}{m_{Tow}}\right)}$$
(62)

301 where γ_{UT} and γ_{AT} are defined to introduce the effect of added mass as well as the rigidity 302 changes at sea level and platform level to the natural mode shapes, respectively. To find the 303 coefficient matrix **D** for each natural mode, Eq. (60) should be solved by substituting the 304 wavenumbers obtained from the singularity of matrix **P**. Finally, the natural mode shapes of the 305 system can be found by substituting variables found for each mode in Eqs. (57), (58) and (59). By 306 merging them, it can be represented in a single function as

$$X_n(z) = X_T(z) H_1(z) + X_{MA}(z)H_2(z) + X_{MU}(z)H_3(z), \qquad 0 \le z \le L$$
(63)

307 3.2. The solution for an external load

308 The solution of an equation of motion in the form of

$$x^{(iv)}(z,t) + A(z) \ddot{x}(z,t) = \frac{1}{EI_{Mon}} H_3(z)Q(z,t)$$
(64)

309 can be obtained by using the expansion theorem to represent the motion of the tower in the form310 of

$$\mathbf{x}(z,t) = \sum_{n=1}^{\infty} X_n(z) T_n(t)$$
(65)

311 where the $X_n(z)$ is the natural mode of the system which satisfy

$$X_n^{(iv)} - \omega_n^2 A(z) X_n(z) = 0$$
(66)

The above equation is obtained by substituting Eqs (50) into (43) and (44) and merging them by using Eq. (9). Substituting Eq. (65) into Eq. (64) yields

$$\sum_{n=1}^{\infty} X_n^{(i\nu)}(z) T_n(t) + A(z) \sum_{n=1}^{\infty} X_n(z) \ddot{T}_n(t) = \frac{1}{EI_{Mon}} H_3(z) Q(z,t)$$
(67)

Multiplying both sides by $X_m(z)$ and integrating over the length of the tower, it yields

$$\sum_{n=1}^{\infty} T_n(t) \int_0^L X_m(z) X_n^{(iv)}(z) dz + \sum_{n=1}^{\infty} \ddot{T}_n(t) \int_0^L A(z) X_m(z) X_n(z) dz$$

$$= \frac{1}{E I_{Mon}} \int_0^L H_3(z) Q(z,t) X_m(z) dz$$
(68)

Substituting $X_n^{(iv)}$ from Eq. (66) into Eq. (68) results in

$$\sum_{n=1}^{\infty} \int_{0}^{L} A(z) X_{m}(z) X_{n}(z) dz \left(\ddot{T}_{n}(t) + \omega_{n}^{2} T_{n}(t) \right) = \frac{1}{E I_{Mon}} \int_{0}^{L} H_{3}(z) Q(z,t) X_{m}(z) dz$$
(69)

316 The natural modes are orthogonal and normalized. Therefore, the above equation is simplified to

$$\ddot{T}_{n}(t) + \omega_{n}^{2}T_{n}(t) = \frac{1}{EI_{Mon}} \int_{0}^{L} H_{3}(z)Q(z,t)X_{m}(z)dz$$
(70)

317 It should be noted that the natural modes are normalized such that

$$\int_{0}^{L} A(z) X_{n}^{2}(z) dz = 1$$
(71)

By finding $T_n(t)$ from solving Eq. (70), it can be substituted into Eq. (65) to obtain the response of the tower.

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- 320 3.3. Solution for the wave load
- 321 The wave load acting on the tower was introduced in section 2.1. It was shown that Eq. (8) governs
- 322 the motion of the tower with the external load in the form of Eq. (13). It is a function of horizontal
- particle wave velocity, u(z, t) which is defined based on the 2^{nd} order wave theory [72] as

$$u(z,t) = f_1 \cosh(kz) \cos(\omega t) + f_2 \cosh(2kz) \cos(2\omega t)$$
(72)

324 where

$$f_1 = \frac{\omega H}{2\sinh(kd)} \tag{73}$$

$$f_2 = \frac{3}{16} \frac{\omega k H^2}{\sinh^4(kd)}$$
(74)

325 Eq. (72) can be rewritten as

$$u(z,t) = f_2 \cosh(2kz) \left(f(z) \cos(\omega t) + \cos(2\omega t) \right)$$
(75)

326 where

$$f(z) = \frac{8 \sinh^3(kd)}{3} \frac{\cosh(kz)}{\cosh(2kz)}$$
(76)

$$Q(z,t) = F_1(z,t) + H_t(z,t) F_2(z,t)$$
(77)

328 where

$$F_1(z,t) = -\rho_w C_M A \omega f_2 \cosh(2kz) \left(f(z) \sin(\omega t) + 2\sin(2\omega t) \right)$$
(78)

$$F_2(z,t) = \frac{1}{2}\rho_w C_D D f_2^2 \cosh^2(2kz) \left(f(z)\cos(\omega t) + \cos(2\omega t)\right)^2$$
(79)

$$H_t(z,t) = \begin{cases} +1, & u(z,t) > 0\\ -1, & u(z,t) < 0 \end{cases}$$
(80)

329 Eq. (77) is the wave load based on Morison's formula rewritten from Eq. (13). $H_t(z, t)$ is a step 330 function representing the absolute value function in his formula. To find the response of an OWT 331 under this load, it needs Eq. (70) to be solved after substituting Eq. (77) in it. The resultant will be 332 an ordinary nonhomogeneous 2nd differential equation. The solution to this differential equation 333 can be found by solving Eq. (70) for $F_1(z,t)$ and $H_t(z,t)$ $F_2(z,t)$ separately and adding them by 334 using the superposition principle. Substituting $F_1(z, t)$ as Q(z, t) into the right side of Eq. (70) and 335 solving the integration with respect to the z variable analytically will lead to the trigonometrical 336 functions depending on the temporal variable left on the right side of Eq. (70). The solution of it 337 is pretty straightforward. So, the solution for $F_1(z, t)$ can be proposed in the form of Eq. (65) 338 where $T_n(t)$ is found by solving Eq. (70) for $F_1(z, t)$.

However, the solution for Eq. (70) when the second term of Eq. (77), $H_t(z,t) F_2(z,t)$, is substituted as Q(z,t) in the right side of it will be challenging. This term consists of $F_2(z,t)$ in Eq. (79) in which temporal and spatial functions are squared and $H_t(z,t)$ in Eq. (80) which is a step function depending on the sign of u(z,t). To obtain a solution for Eq. (70) in this matter, it needs to work on these two parts to transfer them into the conventional form of functions with a combination of spatial functions and linear trigonometrical terms. To start, the squared term of Eq. (79) is expanded to obtain

$$F_{2}(z,t) = \frac{1}{2} \rho_{w} C_{D} D_{Mon} f_{2}^{2} \cosh^{2}(2kz) \left(f^{2}(z) \cos^{2}(\omega t) + 2f(z) \cos(\omega t) \cos(2\omega t) + \cos^{2}(2\omega t)\right)$$
(81)

346 By using the following trigonometrical relationships

$$\cos^{2}(\theta) = \frac{1}{2} (\cos(2\theta) + 1)$$

$$\cos(\theta) \cos(2\theta) = \frac{1}{2} (\cos(\theta) + \cos(3\theta))$$
(82)
(83)

347 and substituting them into Eq. (81), it yields

$$F_{2}(z,t) = \frac{1}{4}\rho_{w}C_{D}D_{Mon}f_{2}^{2}\cosh^{2}(2kz)(f^{2}(z) + 1 + 2f(z)\cos(\omega t) + f^{2}(z)\cos(2\omega t) + 2f(z)\cos(3\omega t) + \cos(4\omega t))$$
(84)

348 which is a function in which the trigonometrical terms are linear.

The value of $H_t(z, t)$ can be determined by the sign of u(z, t). Recalling u(z, t) from Eq. (75) and using the trigonometric relationship in Eq. (82) yields

$$u(z,t) = f_2 \cosh(2kz) \left(2\cos^2(\omega t) + f(z)\cos(\omega t) - 1\right)$$
(85)

351 The above equation reveals that the sign of u(z, t) depends on the values of $\cos(\omega t)$ and f(z).

To evaluate the sign of u(z, t), one needs to find when and where it becomes zero. From Eq. (76),

it can be found that f(z) is continuously decreasing when z is increasing because the numerator is always smaller than the denominator. Also, $\cosh(kz)$ is always positive and $\sinh(kd)$ is positive

is always smaller than the denominator. Also, $\cosh(kz)$ is always positive and $\sinh(kd)$ is positive as long as kd is positive. Therefore, f(z) is always a positive quantity and does not influence the

356 sign of u(z, t). For this reason, the only term that governs the sign of u(z, t) is $f(z) \cos(\omega t) +$

357 $\cos(2\omega t)$. For f(z) > 1, there are two positive roots in the $[0, 2\pi/\omega]$ domain. They are

$$t_1(z) = \frac{1}{\omega} \cos^{-1}(\frac{-f(z) + \sqrt{f^2(z) + 8}}{4})$$
(86)

$$t_2(z) = \frac{2\pi}{\omega} - t_1(z)$$
(87)

358 Therefore,

$$H_t(z,t) = \begin{cases} +1, & 0 \le t < t_1(z) \text{ or } t_2(z) < t \le \frac{2\pi}{\omega} \\ -1, & t_1(z) < t < t_2(z) \end{cases}$$
(88)

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- 359 The above equation represents the value of $H_t(z, t)$ in the time domain of $[0, 2\pi/\omega]$. For the time
- domain beyond it, the value of $H_t(z, t)$ can be evaluated by considering the time variable relative
- to a one-period time frame since $F_2(z, t)$ is a periodic function.
- 362 Therefore, Eq. (13) can be written in the form of

$$Q(z,t) = \sum_{i=1}^{2} P_{1i}(z) \sin(i\omega t) + H_t(z,t) P_2(z) + H_t(z,t) \sum_{j=1}^{4} P_{3j}(z) \cos(j\omega t)$$
(89)

363 where

$$P_{11}(z) = -\rho_w C_M A_{Mon} \omega f_2 \cosh(2kz) f(z)$$
(90)

$$P_{12}(z) = -2\rho_w C_M A_{Mon} \omega f_2 \cosh(2kz)$$
(91)

$$P_2(z) = 0.25 \,\rho_w C_D D_{Mon} f_2^2 \cosh^2(2kz) \left(f^2(z) + 1\right) \tag{92}$$

$$P_{31}(z) = 0.25 \,\rho_w C_D D_{Mon} f_2^2 \cosh^2(2kz) \,2f(z) \tag{93}$$

$$P_{32}(z) = 0.25 \,\rho_w C_D D_{Mon} f_2^2 \cosh^2(2kz) f^2(z) \tag{94}$$

$$P_{33}(z) = P_{31}(z) \tag{95}$$

$$P_{34}(z) = 0.25 \,\rho_w C_D D_{Mon} f_2^2 \cosh^2(2kz) \tag{96}$$

The solution for Eq. (89) can be found by using the superposition principle since the properties of the system is linear. The solution for Eq. (89) can be expanded in the natural modes as follows:

$$x(z,t) = \sum_{n=1}^{\infty} T_n(t) X_n(z)$$
(97)

366 where $T_n(t)$ satisfies

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$$\ddot{T}_n(t) + \omega_n^2 T_n(t) = \sum_{i=1}^2 v_{1ni} \sin(i\omega t) + v_{2n} + \sum_{j=1}^4 v_{3nj} \cos(j\omega t)$$
(98)

$$v_{1ni} = \frac{1}{EI_{Mon}} \int_0^d P_{1i}(z) X_n(z) dz$$
(99)

$$v_{2n} = \frac{1}{EI_{Mon}} \int_0^d H_t(z,t) P_2(z) X_n(z) dz$$
(100)

$$v_{3nj} = \frac{1}{EI_{Mon}} \int_0^d H_t(z,t) P_{3j}(z) X_n(z) dz$$
(101)

367 Solving Eq. (98) is possible by using the superposition principle in three parts namely $T_{1n}(t)$,

368 $T_{2n}(t)$, and $T_{3n}(t)$ where

$$\ddot{T}_{1n}(t) + \omega_n^2 T_{1n}(t) = \sum_{i=1}^2 v_{1ni} \sin(i\omega t)$$
(102)

$$\ddot{T}_{2n}(t) + \omega_n^2 T_{2n}(t) = v_{2n}$$
(103)

$$\ddot{T}_{3n}(t) + \omega_n^2 T_{3n}(t) = \sum_{j=1}^4 v_{3nj} \cos(j\omega t)$$
(104)

369 Therefore,

$$T_n(t) = T_{1n}(t) + T_{2n}(t) + T_{3n}(t)$$
(105)

370 For Eq. (100), substituting into Eq. (103) results in

$$\ddot{T}_{2n}(t) + \omega_n^2 T_{2n}(t) = \frac{1}{EI_{Mon}} \int_0^d H_t(z,t) P_2(z) X_n(z) dz$$
(106)

371 The solution for the above equation can be represented as

$$T_{2n}(t) = \frac{1}{\omega_n} \frac{1}{EI_{Mon}} \int_0^t \int_0^d H_t(z,\tau) P_2(z) X_n(z) \sin(\omega_n(t-\tau)) dz d\tau$$
(107)

372 In the above equation, the function $H_t(z,t)$ poses challenges in solving the integration 373 analytically. To solve the double integration in the above equation, it requires the removal 374 $H_t(z, t)$ from inside the integrations to have a conventional double integration. As defined in Eq. 375 (88), $H_t(z, t)$ governs the sign of v_{2n} . It can be either positive or negative, depending on which 376 time frame it is evaluated. $H_t(z, t)$ is positive in the time frame of $[0, t_1(z)]$, negative in the time 377 frame of $[t_1(z), t_2(z)]$, positive in the time frame of $[t_2(z), t_2(z) + 2t_1(z)]$ and so on. By 378 introducing a new integer variable, m, by which $H_t(z,t)$ is positive when m is an odd integer and 379 an even number when it is negative. Therefore, $H_t(z, t)$ can be redefined as

$$H_t(z,t) = \begin{cases} +1, & m = 1, 3, 5, \dots \\ -1, & m = 2, 4, 6, \dots \end{cases}$$
(108)

380 Since $t \in [t_{e(m-1)}(z), t_{em}(z)]$, one can conclude

$$H_t(z,t) = (-1)^{m-1}$$
(109)

381 where $t_{e0}(z) = 0$, $t_{e1}(z) = t_1(z)$, $t_{e2}(z) = t_2(z)$, $t_{e3}(z) = t_2(z) + 2t_1(z)$, and so on. 382 Substituting the above equation into Eq. (107) and reversing the integration order yields

$$T_{2n}(t) = \frac{1}{\omega_n} \frac{1}{EI_{Mon}} \int_0^d P_2(z) X_n(z) \left(\int_0^{t_1(z)} \sin(\omega_n(t-\tau)) d\tau - \int_{t_1(z)}^{t_2(z)} \sin(\omega_n(t-\tau)) d\tau + \int_{t_2(z)}^{t_2(z)+2t_1(z)} \sin(\omega_n(t-\tau)) d\tau - \cdots \right) d\tau + (-1)^{m-1} \int_{t_{e(m-1)}(z)}^t \sin(\omega_n(t-\tau)) d\tau d\tau$$
(110)

383 Similarly, for Eq. (104), the solution is

$$T_{3n}(t) = \sum_{j=1}^{4} \frac{1}{\omega_n} \frac{1}{EI_{Mon}} \int_0^d P_{3j}(z) X_n(z) \left(\int_0^{t_1(z)} \cos(j\omega t) \sin(\omega_n(t-\tau)) d\tau - \int_{t_1(z)}^{t_2(z)} \cos(j\omega t) \sin(\omega_n(t-\tau)) d\tau + \int_{t_2(z)}^{t_2(z)+2t_1(z)} \cos(j\omega t) \sin(\omega_n(t-\tau)) d\tau - \cdots + (-1)^{m-1} \int_{t_e(m-1)(z)}^t \cos(j\omega t) \sin(\omega_n(t-\tau)) d\tau \right) dz$$
(111)

Therefore, a method of solving the double integration in Eq. (107) is proposed by removing the function $H_t(z,t)$, or better to say the staging of the integration domain, as represented in Eqs. (110) and (111).

387 4. Parametric study and numerical example

388 In this section, the effect of the three solution variables, α_3 , α_5 , and α_6 introduced in Eqs. (52), 389 (55), and (56) representing foundation rotational stiffness, nacelle rotational mass, and nacelle 390 mass, respectively, as well as the water depth on the natural wavenumbers, and the effect of 391 added mass on the response are investigated.

A numerical example is represented for a reference OWT. The geometry of the system has been chosen from the DTU 10 MW three-bladed OWT presented by Bak et al. [74]. The structural properties are summarized in Table 2. Note that the density is considered approximately 8% more than the regular steel density to take into account the mass of the components such as paint, bolts, flanges and stiffeners [75]. The average tower diameter is the average of the tower diameter along its height, and the average thickness is calculated from the actual tower mass [57].

- Table 3 represents the hydrodynamic loading parameters used in this study. The coefficients arechosen by the recommendations provided by DNV-RP-C205 [72].
- 400 The values for the coupled springs model are provided in Table 4 by the work presented by
- 401 Alkhoury et al. [60]. They calculated these values for the loose sand from the finite element model
- 402 created for their study in which the same DTU 10 MW OWT is modeled and studied.

For this case, the solution variables are calculated and represented in Table 5. By using the procedure described in section 3.1, the natural frequencies of the tower, f_n , are calculated for the first six modes and represented in Table 4. The corresponding mode shapes are illustrated in Figure 2 which is to satisfy equation (71).

407 In Table 6, the natural frequencies are calculated for both cases of considering the added mass 408 and without the effect of added mass. As represented, including the effect of added mass in the 409 system decreases the natural frequencies. This effect in the 1st mode is not as significant as in 410 higher modes. The reason can be explained by using Figure 2 where the displacement of the 411 underwater section in 1st mode is remarkably less compared to the higher modes. Besides, the 412 presence of the weighty nacelle-rotor assembly mass dominates the motion of the 1st mode. 413 Therefore, the motion of the system in the higher modes is less than in the 1st mode due to the 414 substantial inertial force at the top of the system.

415 4.1. Comparison with the finite element model and the degree of accuracy

416 To evaluate the accuracy of the results, they are compared with the study conducted by Alkhoury 417 at el. [48,60]. They created a detailed 3D finite element (FE) model within Abagus/Standard to 418 compute the natural frequencies of the DTU 10 MW OWT. They used shell elements to model the 419 tower, including the diameter variation in length and solid elements for monopile. They also fully 420 modeled the soil inside and outside the monopile to investigate the soil structure interaction. 421 They also performed a parametric study on the 1st natural frequency by varying the water depth 422 and monopile's diameter and thickness. The 1st natural frequencies of the system are calculated 423 and compared with the values they calculated for the loose sand that are represented in Table 7. 424 Note that the values of the coupled springs used in this study are also calculated by them which 425 are obtained from the FE model. The differences between the results obtained by the proposed 426 model and the FE model reveal that the proposed model underestimated the 1st natural frequency 427 for every water depth in the range between 13% to 16.8%. Alkhoury et al. [60] also compared the 428 results of the full 3D model with the one in which the tower cross-section is constant for a water 429 depth of 25 m. They found that simplifying modeling by considering the tower's cross-section 430 constant reduced the 1st natural frequency by 11% for the monopile with 8.3 m in diameter and 431 9 cm in thickness. The findings of this paper also verify this underestimation with a 13.8% 432 deviation. Therefore, this simplification underestimates the 1st natural frequency that requires 433 using more complicated equations of motion to improve the accuracy of the natural frequency 434 estimation.

435 4.2. Parametric study on natural wavenumbers

436 The effect of the water depth and solution parameters, α_3 , α_5 , and α_6 , on the tower 437 wavenumber, β_{Tn} , have been parametrically studied for the first 5 modes and the results have 438 been represented in Figure 3 to Figure 6. In these figures, the tower wavenumbers are normalized 439 to the values of β_{Tn} when $\alpha_3 = \alpha_5 = \alpha_6 = d = 0$. In this study, the ratio of water depth and 440 tower length, d/L, varies from 0 to 1, implying d = 0 and d = L, respectively. The variation of 441 the solution variables as well as their corresponding variation of the parameters used in the 442 parametric study, is represented in Table 8. It should be noted that the effect of the support's 443 lateral stiffness, K_L , and cross-coupled stiffness, K_{LR} , which are used in variables α_1 and α_2 ,

respectively, are well investigated in the literature in ref. [56]. So, their effect is not included in

the parametric study. Besides, the section of the system is kept constant throughout the lengthfor simplicity.

447 4.2.1. The effect of water depth

448 As illustrated in Figure 3, the wavenumber for the first 5 modes decreases by increasing the water 449 depth regardless of the boundary conditions properties because of the presence of added mass 450 to the system for the case of the properties introduced in Table 8. The value of the wavenumber 451 in the 1st mode remains almost constant by increasing the water depth up to 0.4L and drops up 452 to L for the case of $\alpha_6 = 0$ while the variation is almost constant for the case when $\alpha_6 = 10^{-6}$. 453 This can be because of the domination of the heavy mass in the motion of the system in the 1st 454 mode. The decrease of the wavenumber for the 2nd mode starts at 0.2L and decreases 455 approximately the same amount as the 1st mode at the sea level equal to the tower length for the 456 cases when $\alpha_6 = 0$. The initiation of the drop of wavenumber for higher modes is almost half of 457 the previous modes. It can be concluded that the effect of shallow water compared to the tower 458 length and consequently the added mass in the lower modes is not significant as opposed to 459 higher modes where it drops immediately by increasing sea level. This phenomenon may be 460 important by the fact that the free sea level varies in each wave period. Therefore, the water 461 surface variation can significantly change higher natural wavenumbers in the shallow water 462 proportional to the tower length for each period of wave load. However, for higher values of d/L, 463 the lower natural modes are also influenced by sea level variation. This variation of the natural 464 frequencies is also reported in the literature for the 1st natural modes based on the measured 465 data in refs. [61,65].

The pattern of the variation of wavenumbers by varying the water depth shown in Figure 3 reveals a wavy-shape decrease in which the reduction rate changes in different mode numbers. It can be seen that the number of crests in this pattern is equal to the mode number. For instance, for the 2^{nd} mode, two crests at around d/L = 0.2 and d/L = 0.8 can be seen while 5 distinguished crests are visible in the figure for mode 5. Therefore, it can be concluded that the variation of wavenumbers versus d/L is converging to a linear reduction rate.

472 4.2.2. The effect of support rotational stiffness, α_3

473 Figure 4 illustrates the variation of the wavenumber against the different values of α_3 in 1st and 5th modes at d/L = 0.227, for instance. As expected, by increasing α_3 , or decreasing the 474 475 support's rotational stiffness, the value of the wavenumber decreases linearly in the 1st mode and 476 nonlinearly in the 5th mode. By increasing the rotational softness of the support the wavenumbers 477 for all natural modes decrease. This is because the higher rotational softness provides higher 478 rotation in the support resulting in a reduction of the wavenumbers. In the 1st mode, the effect of 479 α_3 is more than in the 5th mode. Also, the impact of the α_3 is less in higher values of the α_5 and 480 α₆.

481 4.2.3. The effect of nacelle-rotor assembly rotational moment of inertia, α_5

482 The parametric study on α_5 has been illustrated in Figure 5 for the 1st and 5th modes for two values 483 of $\alpha_3 = \infty \& 0.2$ and $\alpha_3 = 0 \& 10^{-6}$ for d/L = 0.6 when the values of α_5 varying between 0 and

484 10^{-3} . As it can be seen in Figure 5-a, the wavenumber decreases by increasing α_5 for the 1st mode. 485 For the 6th mode, the natural wavenumber is almost insensitive to the variation of α_5 despite a 486 rapid decrease of natural wavenumber at small values of α_5 . The decrease of wavenumber can 487 be explained by the fact that the rotational moment of inertia at the top of the tower increases 488 the mass momentum of the system at the top. Therefore, the moment of inertia due to the 489 nacelle-rotor assembly causes the system to oscillate slower, yielding to the lower values of the 490 natural wavenumbers.

491 4.2.4. The effect of the nacelle mass, α_6

492 Figure 6 illustrates the variation of the wavenumber against α_6 ranging from 0 to 10^{-6} for two 493 values of $\alpha_3 = \infty \& 0.2$ and $\alpha_5 = 0 \& 10^{-3}$ in 1st and 5th mode when d/L = 0.6. A reduction can 494 be seen in Figure 6-a for all cases of α_3 and α_5 in 1st mode. By increasing α_6 , which is the increase 495 of the top mass with respect to the tower's flexural rigidity, the motion of the tower becomes 496 slower yielding to the smaller values of the natural wavenumbers.

497 The reduction effect of α_6 , which is proportional to the nacelle-rotor assembly mass, can be 498 explained by the whipping effect of the tower. The inertia of the heavy mass at the top of the 499 tower may cause a delayed motion relative to the mid-section of the tower in the same direction. 500 When the mid-section of the tower reaches its maximum displacement, the top section is still 501 moving imposing extra shear force to the mid-section pushing it to move further, consequently, 502 increasing the oscillation period, decreasing the frequency, and decreasing the wavenumber of 503 the system, see Eq. (50). Physically speaking, the heavy mass at the top of the tower produces an 504 inertia force in the opposite direction of the motion which slows down the motion of the tower.

505 In higher modes, as illustrated in Figure 6, the variation of the natural wavenumber of the system 506 is a smooth reduction despite a rapid reduction in the small values of α_6 . The reason for that can 507 be explained by the fact that the translational acceleration at the top of the tower in higher modes 508 is relatively smaller than that of the lower modes due to the whipping effect explained earlier. 509 This makes higher modes less sensitive to the variation of the nacelle mass. Furthermore, this 510 effect can also be seen in Figure 3 where the curves are gathering together by increasing the mode 511 number implying that the effect of the boundary conditions at both sides of the system are fading 512 out of the natural wavenumbers.

513 4.3. The response of the reference tower to the wave load

The wave load introduced in Eq. (89) is applied to the system and the response is evaluated by Eq. (97). The properties of Stokes's wave kinematics applied to Morison's formula are briefly represented. The proposed model's ability to deal with the difficulties posed by the drag term is explained. Besides, the effect of the added mass on the response is investigated for a certain wave load configuration. Finally, a comparison is performed between the responses obtained by the proposed solution and the numerical one.

- 520 4.3.1. The application of the wave load in the proposed solution
- 521 The inherent properties of wave load based on Morison's formula with Stokes's wave kinematics 522 raise some difficulties in evaluating the response of the tower. Before presenting how the

523 proposed solution deals with these properties, one needs to discuss the properties of the wave 524 load with Morison's equation introduced in Eq. (89). To this end, an ocean wave with a height of 525 5.1 m and a length of 132 m is chosen. This ocean wave represents a normal sea state of a water 526 depth of 35 m in a wind speed of 26 m/s which is reported as a nonlinear ocean wave state in the 527 ref. [76]. The properties of this ocean wave are represented in Table 9. The wave horizontal 528 particle velocity, u(z, t), introduced in Eq. (72) and the corresponding ocean wave load are drawn 529 in Figure 7 a&b, respectively. As expected, the wave horizontal particle velocity is a periodic but 530 non-symmetric function, which is the property of the 2nd order wave theory. The wave load is also 531 not started from zero as shown in Figure 7 because of the presence of the cosine function in the 532 drag term of the wave load formula.

533 Moreover, Figure 7-a reveals that the wave horizontal particle velocity, u(z, t), does not become

zero at the same time for all values of z. This is the second property of the 2nd order wave theory that causes difficulties in obtaining the response of a system loaded with it. Therefore, the situation in which u(z,t) = 0, depends on temporal and special variables. This is shown more precisely in Figure 8 for the variation of u(z,t) in the z-direction for some instant of time around

538 the first zero value. Therefore, two functions of $t_1(z)$ and $t_2(z)$ represented by Eqs. (86) and (87) 539 are defined to evaluate the time when u(z,t) is zero. The importance of defining these two 540 functions is to evaluate the absolute value function in the drag term of the wave load, 541 u(z,t)|u(z,t)|, in Eq. (13). To compensate for the absolute value function, $H_t(z,t)$ is defined as 542 a function $t_1(z)$ and $t_2(z)$ in Eq. (80). This poses difficulties in evaluating Eqs. (100) and (101). By 543 defining $H_t(z,t)$ as Eq. (109) and carrying out the integration, it becomes possible to remove 544 $H_t(z, t)$. In addition, changing the integration order by which the temporal integration is taken 545 first in the domain as a function of $t_1(z)$ and $t_2(z)$ results in trigonometrical functions. By 546 introducing $t_1(z)$ and $t_2(z)$ and their combinations in periods, the special outer integration can 547 be evaluated analytically since those are functions of reversed trigonometrical functions as can 548 be seen in Eqs. (86) and (87). Therefore, the response can be obtained as a function without any 549 need for numerical evaluation.

The presence of $H_t(z, t)$ in the wave load is quite essential. To show this, an imaginary wave load based on Eq. (77) is defined as $F_1(z, t) + F_2(z, t)$. By drawing it together with the wave load in Eq. (77), the effect of the $H_t(z, t)$ reveals itself. Figure 9 illustrates this comparison at the sea level, z=35. It can be seen that $H_t(z, t)$ causes significant changes in the wave load when u(z, t)is negative between $t_1(z)$ and $t_2(z)$. Therefore, its presence cannot be neglected in evaluating the response. However, the severity of this effect may be different in other wave configurations.

556 4.3.2. The response of the reference tower to the 2nd wave load

557 The response history of the system is illustrated in Figure 10 at different heights in which the first 558 5 natural modes are participating. The selected wave height and length correspond to a wave 559 frequency of 0.1 Hz, T= 9.52 sec, which is way below the first natural frequency of the reference 560 tower, 0.1663 Hz, represented in Table 9. As seen in Figure 10, the maximum deflection occurs at 561 the top of the towers reaching 0.13 m. In addition, the time history deflection curve in Figure 10 562 reveals that the responses are a non-periodic vibration even though the loading is periodic. This 563 can be because of the indirect presence of the term $H_t(z,t)$ in the response in Eqs. (110) and 564 (111). The indirect presence of $H_t(z, t)$ in the response shows itself by being displayed in different time frames when the direction of the wave horizontal particle velocity changes. This triggers the transient responses at the beginning of each stage causing the response to be non-periodic. Moreover, Figure 10 shows that the natural frequencies originating from the transient response are carried by the steady-state response. This also causes the response to be non-periodic.

569 The deflection of the reference tower at the early stages of motion is illustrated in Figure 11. The 570 motion starts from the rest initial condition and follows by imposing the wave load up to sea level, 571 in this case, 35 m, causing the lower sections of the tower to move while the movement of the 572 upper section is delayed because of the nacelle mass and tower softness. This can be referred to 573 as the whipping effect when the motion of the upper sections is amplified by the motion of the 574 lower sections of the tower. This effect can be seen clearly in the motion of the reference tower 575 which has low stiffness or heavy nacelle mass at the top. Moreover, by looking at the deflection 576 of the reference tower in Figure 11 for time instants from t/T=0.08 up to 0.1, T being the period 577 of the wave which is 9.6 sec, it implies that the motion of the tower in low sections is slowing 578 down while the upper section is still moving towards the negative deflections representing an 579 instance for the whipping effect. It should be mentioned that the wave load at zero time is a 580 positive value, as seen in Figure 7-b, decreasing in the early stages of the loading and reversing its 581 direction as time passes.

582 4.3.3. The effect of added mass on the response

583 The inertia term of Morison's formula adds an extra mass to the system up to the sea level. To 584 represent its effect on the response of the reference tower, it is evaluated by considering $C_A = 1$ 585 and $C_A = 0$ to simulate a system with and without the presence of the added mass, respectively, 586 and the time history responses are illustrated in Figure 12 at the nacelle level. As mentioned in 587 section 4.1.1 and illustrated in Figure 3, the value of the system's natural wavenumbers decreases 588 by increasing the sea level resulting from the decrease in natural frequencies based on Eq. (50). 589 This leads the system with added mass oscillating with higher natural periods falling forward than 590 the one without added mass, as seen in Figure 12. Furthermore, as seen in Table 6, the added 591 mass is significantly influenced in the 2nd and higher natural frequencies in the case of the system 592 of this study. The differences between the natural frequencies in higher modes are visible in 593 Figure 12.

594 The effect of the added mass up to the sea level can also be seen in the early stages of the loading 595 in Figure 13. The added mass increases the inertia force of the system. The higher the inertia force 596 is, the slower the vibration results. Therefore, the system's motion with added mass included 597 delays compared to the one without the added mass as seen in Figure 13. It is also revealing that 598 the added mass is appended to the system up to sea level, providing lower deflection at the lower 599 sections of the system while the upper sections have almost the same deflection. The overall 600 interpretation from Figure 13 reveals that the effect of the added mass up to the sea level is 601 successfully simulated by the proposed solution.

602 4.3.4. Comparison with numerical results

The derived formulation of the response of the system is based on the expansion in the natural
 modes. The accuracy of the solution depends on the number of natural modes participating in the
 solution. For a continuous system, an infinite number of modes is expected. The higher number

606 of modes participating in the solution, the more accurate the response obtained. It is worthwhile 607 to mention that the Bernoulli-Euler beam theory in the form of Eq. (1) is applicable for the lower 608 natural modes only. In higher modes, the shearing deformation and rotatory inertia significantly 609 affect the natural frequencies [50]. Therefore, reaching higher natural modes requires the 610 equation of motion with those effects considered. However, as far as the wave load by using 611 Morison's formula matters, its validity requires that the ocean wavelength should be five times 612 [72] higher than the diameter of the slender structure. This limitation causes the frequency of the 613 ocean wave load to be lower than the system's first natural frequency. Therefore, only the first 614 few natural modes may be enough to evaluate the tower's response.

- To confirm the accuracy of the proposed solution, a numerical evaluation is performed by using
 the standard commercial software *Mathematica*[®] [77]. The partial differential numerical solver, *NDSolve*, is chosen to solve Eq. (8) by introducing the wave velocity from Eq. (72). The properties
- of the system chosen for the numerical evaluation are represented in Table 10.

619 The wave height, H, and length, λ , are selected to be 3 and 100 meters, respectively. By setting 620 MaxStepSize equal to 1.4, and AccuracyGoal and PrecisionGoal to 6, NDsolve solves the PDE by 621 using the Hermite method in orders of 7 and 3 in z and t variables, respectively, in the domain of 622 $z \in [0,115]$ and $t \in [0,34]$. Besides, the first 6 natural modes of the tower with the properties 623 and loading the same as the numerical one are selected for participating in the proposed solution 624 results. The comparison between the response obtained by the numerical solution and the 625 proposed one verifies the perfect agreement between the two methods, as illustrated in Figure 626 14.

627 5. Conclusions

628 An analytical solution for the modal analysis of offshore wind turbine strucutres has been 629 developed. The solution includes the wave-structure interaction by appending an extra mass to 630 the system underwater. In an effort to propose a more accurate solution based on the classical 631 analytical methods, the flexibility of the foundation as well as the inertial forces induced by the 632 nacelle-rotor assembly translational and rotational inertia are assigned to the boundaries of the 633 system. Besides, the considerable cross-sectional changes at the platform level where the 634 monopile is connected to the tower by a transition part are taken into account in the solution. 635 Overall, a system of three partial differential equations consisting of 12 boundary conditions and 636 2 initial conditions has been solved using the expansion theorem.

The effect of water depth, foundation rotational flexibility, nacelle mass, and nacelle-blades
rotational inertia on the system's natural wavenumber were studied parametrically for the first 5
natural modes of the system. The results reveal that:

6401. The natural wavenumber decreases by increasing the water depth to the tower-length641ratio, d/L, for all natural modes producing a wavy pattern based on the modal number.642The effect of the foundation rotational flexibility, nacelle-blades rotational inertia, and643nacelle mass on the natural wavenumber decreases by increasing the modal number for644all sea level values. More importantly, the variation of the natural wavenumbers by645variation of the sea level implies that the natural wavenumbers or natural frequencies of

- the system vary during sea level variation during a period of wave load. Therefore, the
 system's natural frequencies can be considered a time-dependent quantity. It can be
 essential in the assessment of the ringing-type resonance of the system and fatigue
 loading.
- 650 2. The system's natural frequencies decrease by increasing the foundation rotational
 651 flexibility, mass and the rotational inertial of the nacelle-rotor assembly. This pattern has
 652 been seen at all water depths.
- 3. The proposed model, based on the simplification of considering the constant cross-section for the tower, underestimates the 1st natural frequency of the system between
 13% to 16.8%. Reaching higher accuracy requires establishing more complicated
 equations of motions by accounting for the cross-sectional variation of the tower.
 However, the proposed method is straightforward and agile in calculating cost-efficient
 natural frequencies.
- The solution for the undamped response of the tower under the wave load with 2nd order Stokes's
 wave theory based on Morison's formula has been developed as an analytical function. Two major
 contributions are
- 6621. The drag term of Morison's formula, neglected by many researchers, is successfully663included in finding the response of the system by defining $H_t(z,t)$ to remove the absolute664value function in the drag term.
- Constraints
 2. The comparison made between the responses of the system with and without added
 mass showed that the presence of added mass up to the sea level changes the shape of
 the response. This also can be important in fatigue evaluation of the system by providing
 a more realistic estimation of the stress status in the structure.

669 6. References

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,	Structural properties
L	Nacelle level from the seabed (m)
L_{Tow}	Tower length (m)
D_{Tow}	Tower average diameter (m)
t_{Tow}	Tower average thickness (m)
m_{Tow}	Tower mass of unit length (k g/m)
E_{Tow}	Tower Young's modulus (GPa)
EI_{Tow}	Flexural rigidity of the tower, i.e., $E_{Tow}I_{Tow}$ (GPa. m^4)
M_n	Nacelle-Rotor assembly mass (k g)
J_p	Nacelle-Rotor assembly rotational inertia (k $g.m^2$)
L_{Plat}	Platform level from the seabed (<i>m</i>)
D _{Mon}	Monopile average diameter (m)
t _{Mon}	Monopile average thickness (m)
4	Mononile cross-sectional area (m^2)
A _{Mon}	
A _{Mon} m _{Mon}	Monopile mass of unit length (kg/m)
A _{Mon} m _{Mon} E _{Mon}	Monopile mass of unit length (kg/m) Monopile Young's modulus (<i>GPa</i>)
A _{Mon} m _{Mon} E _{Mon} EI _{Mon}	Monopile mass of unit length (k g/m) Monopile Young's modulus (GPa) Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)
$\frac{A_{Mon}}{m_{Mon}}$ $\frac{E_{Mon}}{EI_{Mon}}$ ρ_s	Monopile closs sectional area (m^2) Monopile mass of unit length (kg/m) Monopile Young's modulus (GPa) Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ $(GPa.m^4)$ Material Density (kg/m^3)
$\frac{A_{Mon}}{m_{Mon}}$ $\frac{E_{Mon}}{EI_{Mon}}$ $\frac{\rho_s}{Symbol}$	Monopile closs sectional area (m^2) Monopile mass of unit length (kg/m) Monopile Young's modulus (GPa) Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)Material Density (kg/m^3) Support Stiffness
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline E I_{Mon} \\ \hline \rho_s \\ \hline \textbf{Symbol} \\ \hline K_L \end{array}$	Monopile closs sectional area $(m^2)^2$ Monopile mass of unit length (kg/m) Monopile Young's modulus (GPa) Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ $(GPa.m^4)$ Material Density (kg/m^3) Support StiffnessLateral stiffness (GN/m)
$\begin{array}{c} A_{Mon} \\ m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_s \\ \hline \textbf{Symbol} \\ \hline K_L \\ \hline K_{LR} \end{array}$	Monopile closs sectional area $(m^2)^2$ Monopile mass of unit length (kg/m) Monopile Young's modulus (GPa) Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)Material Density (kg/m^3) Support StiffnessLateral stiffness (GN/m) Cross stiffness (GN)
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_s \\ \hline Symbol \\ \hline K_L \\ \hline K_{LR} \\ \hline K_R \end{array}$	Monopile closs sectional area (m^{2})Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^{4}$)Material Density (kg/m^{3})Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_{s} \\ \hline \\ $	Monopile closs sectional area (m^{2})Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^{4}$)Material Density (kg/m^{3})Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)Hydrodynamic loading properties
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_s \\ \hline Symbol \\ \hline K_L \\ \hline K_{LR} \\ \hline K_R \\ \hline \\ Symbol \\ \hline d \end{array}$	Monopile closs sectional area (m)Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)Material Density (kg/m^3)Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)Hydrodynamic loading propertiesWater depth (m)
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_{S} \\ \hline \\ \hline Symbol \\ \hline \\ K_{LR} \\ \hline \\ K_{R} \\ \hline \\ \hline \\ Symbol \\ \hline \\ d \\ \hline \\ C_{D} \end{array}$	Monopile closs sectional area (m^{2})Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^{4}$)Material Density (kg/m^{3})Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)Hydrodynamic loading propertiesWater depth (m)Drag coefficient
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_s \\ \hline \\ \hline Symbol \\ \hline K_L \\ \hline K_L \\ \hline \\ K_R \\ \hline \\ \hline \\ Symbol \\ \hline \\ d \\ \hline \\ C_D \\ \hline \\ \hline \\ C_A \end{array}$	Monopile closs sectional area (m)Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)Material Density (kg/m^3)Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)Hydrodynamic loading propertiesWater depth (m)Drag coefficientAdded mass coefficient
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_{S} \\ \hline \\ \hline Symbol \\ \hline \\ K_{LR} \\ \hline \\ K_{R} \\ \hline \\ \hline \\ Symbol \\ \hline \\ \hline \\ C_{D} \\ \hline \\ \hline \\ C_{A} \\ \hline \\ \hline \\ C_{M} \\ \hline \end{array}$	Monopile closs sectional area (m)Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)Material Density (kg/m^3)Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)Hydrodynamic loading propertiesWater depth (m)Drag coefficientAdded mass coefficientInertia coefficient
$\begin{array}{c} A_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_s \\ \hline \\ \hline Symbol \\ \hline \\ K_L \\ \hline \\ K_L \\ \hline \\ K_R \\ \hline \\ \hline \\ Symbol \\ \hline \\ \\ C_D \\ \hline \\ C_A \\ \hline \\ \hline \\ C_M \\ \hline \\ \rho_w \\ \end{array}$	Monopile closs sectional area (m)Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)Material Density (kg/m^3)Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)Hydrodynamic loading propertiesWater depth (m)Drag coefficientAdded mass coefficientInertia coefficientSea water density (kg/m^3)
$\begin{array}{c} A_{Mon} \\ m_{Mon} \\ \hline m_{Mon} \\ \hline E_{Mon} \\ \hline EI_{Mon} \\ \hline \rho_s \\ \hline \\ \hline Symbol \\ \hline \\ K_L \\ \hline \\ K_L \\ \hline \\ K_R \\ \hline \\ \hline \\ Symbol \\ \hline \\ \hline \\ C_D \\ \hline \\ \hline \\ C_A \\ \hline \\ \hline \\ C_M \\ \hline \\ \hline \\ \rho_w \\ \hline \\ \lambda \\ \end{array}$	Monopile closs sectional area (m)Monopile mass of unit length (kg/m)Monopile Young's modulus (GPa)Flexural rigidity of the monopile, i.e., $E_{Mon}I_{Mon}$ ($GPa.m^4$)Material Density (kg/m^3)Support StiffnessLateral stiffness (GN/m)Cross stiffness (GN)Rotational stiffness ($GN.m$)Hydrodynamic loading propertiesWater depth (m)Drag coefficientAdded mass coefficientInertia coefficientSea water density (kg/m^3)Ocean wavelength (m)

Table 1- Symbols definition

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Figure 1- a) A typical OWT configuration, b) Schematic of the model, c) wave loading direction and system coordinates

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	Symbol	Value
Tower length (m)	L _{Tow}	119
Tower average diameter (m)	D_{Tow}	9.6
Tower average thickness (m)	t_{Tow}	0.0295
Tower Young's modulus (GPa)	E _{Tow}	210
Nacelle-Rotor assembly mass (kg)	M _n	676723
Nacelle-Rotor assembly rotational inertia (kg.m2)	J _p	1.7 x 10 ⁸
Platform level from mudline (m)	L_{Plat}	45
Monopile average diameter (m)	D _{Mon}	8.3
Monopile average thickness (m)	t_{Mon}	0.09
Monopile Young's modulus (GPa)	E _{Mon}	210
Material density (kg/m3)	ρ_s	8500

Table 2- DTU 10 MW OWT structural properties [74]

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Table 3- Hydrodynamic loading properties

1	, ,, ,, ,, ,, ,, ,, ,, ,, ,, ,, ,, ,, ,	
	Symbol	Value
Water depth (m)	d	35
Drag coefficient	C_D	0.65
Added mass coefficient	C_A	1
Inertia coefficient	C_M	2
Sea water density (kg/m3)	ρ_w	1025

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Table 4- The values of the coupled springs [60]

	Symbol	Value
Lateral stiffness (GN/m)	K_L	2.48
Cross stiffness (GN)	K_{LR}	-20.7
Rotational stiffness (GN.m)	K _R	412

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Table 5- Solution variables

d/l	a _{Tow}	a_{MA}	a _{MU}	α ₁	α2	α3	$lpha_4$	α_5	α ₆	γ_{AT}	γ_{UT}
0.227	12125	14585	7501	5.84E-04	-4.88E-03	0.097	0.188	1.59E-04	8.48E-07	0.911	1.271

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 Table 6- The natural mode frequencies of the system with and without the effect of added mass

Mada Numbar	fn (Hz)	Difforence (%)		
	No added mass	Added mass	Difference (%)	
1	0.166561	0.166393	-0.1	
2	1.13463	1.0322	-9.0	
3	2.3888	1.98416	-16.9	
4	4.3686	3.8174	-12.6	
5	8.025	6.593	-17.8	
6	12.198	9.8905	-18.9	

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Figure 2- The normalized natural mode shapes of the system with the added mass (MSL=Mean Sea Level)

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Water	Platform	Monopile outer	Monopile thickness	1st Natural	Deviation (%)	
		diameter (m)	(cm)	Alkhoury et al. [60]	Proposed Model	(70)
			9	0.2009	0.1731	-13.8
		8.3	10	0.202	0.1743	-13.7
	8.3		12	0.2056	0.1771	-13.9
			10	0.2074	0.177	-14.7
25	35	9	12	0.2112	0.1795	-15.0
			14	0.2158	0.1813	-16.0
			11	0.2177	0.1809	-16.9
		10	13	0.2186	0.1828	-16.4
			15	0.2207	0.1837	-16.8
			9	0.1909	0.1663	-12.9
		8.3	10	0.1928	0.1682	-12.8
			12	0.197	0.171	-13.2
	45		10	0.1992	0.1719	-13.7
35		9	12	0.2038	0.1734	-14.9
			14	0.2092	0.176	-15.9
			11	0.2096	0.1768	-15.6
		10	13	0.2129	0.1784	-16.2
			15	0.2155	0.1795	-16.7
			9	0.1811	0.1592	-12.1
		8.3	10	0.1836	0.1614	-12.1
			12	0.189	0.1648	-12.8
			10	0.1909	0.1658	-13.1
45	55	9	12	0.1962	0.1686	-14.1
		_	14	0.2023	0.1708	-15.6
			11	0.203	0.1717	-15.4
		10	13	0.207	0.1736	-16.1
			15	0.2101	0.175	-16.7
6 7 8 9 0 1 2 3 3	40					<u>.</u>

Table 7- Comparison of	of the 1 st natural	frequency b	between the pro	posed method and	full 3D FE-based model by [60]
		j. eque.e. e, ~			

Solution variable	Equivalent to
$\alpha_1=\infty$	$K_L = \infty$
$\alpha_2 = 0$	$K_{LR}=0$
$0.2 < \alpha_3 < \infty$	$271 (GN.m) < K_R < \infty$
$lpha_4=1$	$EI_{Mon} = EI_{Tow} = 1357 \ GN. m^2$
$0 < \alpha_5 < 10^{-3}$	$0 < J_P < 1.35 \times 10^9 Kg. m^2$
$0 < \alpha_6 < 10^{-6}$	$0 < M_n < 1357 ton$
$\gamma_{AT} = 1$	$A_{Mon} = A_{Tow}$
$\gamma_{UT} = 1.36$	See Eq. (62)
Received when the	

Table 8- The range of solution variables used for the parametric study and the equality to the parameters







Figure 4- Variation of normalized wavenumber of the system for the above water section, β_{Tn} , for different values of α_3 in the 1st and 5th modes.



Figure 5- Variation of normalized wavenumber of the system for the above water section, β_{Tn} , for different values of α_5 in the 1st and 5th modes.

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Figure 6- Variation of normalized wavenumber of the system for the above water section, β_{Tn} , for different values of α_6 in the 1st and 5th modes.





Figure 7- a) The wave horizontal particle velocity, and b) wave load obtained from Morison's equation, for H=5.1 m and λ =132m.

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Table 9- Specification	of the ocean wave applied	l to the reference tower

Height	Ocean wavelength	Water depth	Frequency	Period	$\frac{H}{a T^2}$	$\frac{d}{aT^2}$	$\frac{H}{d}$	Ursell Number
H (m)	λ (m)	d (m)	f (Hz)	T (sec)	g I	y I	u	Ur
5.1	132	35	0.105	9.53	0.00572	0.0392	0.145	2.072

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Figure 8- The variation of the wave horizontal particle velocity at around $t_1(z)$, for wave with H = 5.1 m and $\lambda =$ 132 m.

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Figure 9- The effect of the term $H_t(z,t)$ in Morison's formula with the 2^{nd} order wave kinematics for H = 5.1 m and $\lambda = 132 \ m \ at \ z=35 \ m.$

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Figure 11- The response of the reference tower at the early stages of the motion.



Figure 12- Comparison between the response of the tower at the hub level with and without the added mass.



Figure 13- Comparison between the response of the reference tower with and without the added mass at early stages of the motion

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Symbol Symbol Structural properties Value Support Stiffness Value Lateral stiffness (GN/m) Tower length (m) 70 K_L L_{Tow} ∞ D_{Tow} 6 Cross stiffness (GN) Tower average diameter (m) K_{LR} 0 Tower average thickness (m) 0.045 Rotational stiffness (GN.m) K_R 205.72 t_{Tow} Tower Young's modulus (GPa) 210 E_{Tow} Hydrodynamic loading Nacelle-Rotor assembly mass (kg) 0 Symbol Value M_n properties Nacelle-Rotor assembly rotational d 0 Water depth (m) 30 J_p inertia (kg.m2) L_{Plat} Platform level from mudline (m) 45 Drag coefficient C_D 0.65 Monopile average diameter (m) 6 Added mass coefficient D_{Mon} C_A 1 Monopile average thickness (m) 2 t_{Mon} 0.045 Inertia coefficient C_M Monopile Young's modulus (GPa) 210 1020 Sea water density (kg/m3) E_{Mon} ρ_w Material Density (kg/m3) 7820 ρ_s

Table 10-The properties of an OWT for the numerical comparison

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Figure 14- Comparison of the results of the proposed solution with the numerical one at z=115 m.

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